#### Lecture on

New approaches to gradient-based optimization for modelling and control of nonlinear systems of uncertain dynamics:

applications to robotics and electric power generation

**Gerasimos Rigatos** 

Unit of Industrial Automation Industrial Systems Institute 26504, Rion Patras, Greece

email: grigat@ieee.org

#### 1. Outline

• The functioning of **nonlinear dynamical systems** in real conditions is characterized by **model uncertainty**, **parametric changes** and **external perturbations**.

• Control schemes must perform **simultaneously identification** and **stabilization** of such uncertain dynamics.

• This is a dual optimization problem since modelling errors and deviation of the system's state vector elements from the associated setpoints have to be minimized in real-time.

• To achieve these objectives an initial **transformation** (diffeomorphism) of the system's dynamic model to an **equivalent linearized form**, is proposed.

• The transformed control inputs consist of unknown nonlinear functions which are identified with the use of nonlinear regressors.



• Learning in such networks is performed through gradient algorithms in which the adaptation rate (step for the search of an optimum) is defined by conditions for the minimization of an aggregate energy function (Lyapunov function).

#### 1. Outline

• In each iteration of the control algorithm, the estimates of the nonlinear functions that constitute the system's dynamics are fed into a state feedback controller.

• It has been proven that this control approach assures the **minimization of the aforementioned energy function** and thus the nonlinear system becomes **a globally asymptotically stable one**.

• The proposed method can be applied to **all dynamical systems** which satisfy the **differential flatness property**.

• This is the **widest class of nonlinear dynamical systems** to which one can apply **optimization and control with gradient methods**, while assuring the convergence of the optimization procedure and the stability of the control loop.

• The efficiency of the proposed **optimization-based modelling and control approach** has been confirmed in several test cases, concerning complex nonlinear dynamical systems

• In particular, the method has been applied to several electromechanical systems, including robotic systems and electric power generation systems



#### 2. Differential flatness of MIMO nonlinear systems

- Differential flatness theory has been developed as a global linearization control method by M. Fliess (Ecole Polytechnique, France) and co-researchers (Lévine, Rouchon, Mounier, Rudolph, Petit, Martin, Zhu, Sira-Ramirez et. al)
- A dynamical system can be written in the ODE form  $S_i(w, w, w, ..., w^{(i)})$ , i = 1, 2, ..., qwhere  $w^{(i)}$  stands for the i-th derivative of either a state vector element or of a control input
- The system is said to be differentially flat with respect to the flat output

 $y_i = \phi(w, w, w, ..., w^{(a)}), i = 1, ..., m$  where  $y = (y_1, y_2, ..., y_m)$ 

if the following two conditions are satisfied

(i) There does not exist any differential relation of the form

$$R(y, y, y, ..., y^{(\beta)}) = 0$$

which means that the flat output and its derivatives are linearly independent

(ii) All system variables are **functions of the flat output and its derivatives** 

$$v^{(i)} = \psi(y, y, y, ..., y^{(\gamma_i)})$$



#### 2. Differential flatness of MIMO nonlinear systems

The proposed optimization-based control method is based on the transformation of the nonlinear system's model into the linear canonical form, and this transformation is succeeded by exploiting the system's differential flatness properties

• All single input nonlinear systems are differentially flat and can be transformed into the linear canonical form

One has to define also which are the **MIMO nonlinear systems** which are differentially flat.

- Differential flatness holds for **MIMO nonlinear systems** that admit **static feedback linearization** and which can be transformed into the linear canonical form through a change of variables (diffeomorphism) and feedback of the state vector.
- Differential flatness holds for **MIMO nonlinear models** that admit **dynamic feedback** linearization, This is the case of specific underactuated robotic models. In the latter case the state vector of the system is extended by considering as additional flat outputs some of the control inputs and their derivatives
- Finally, a more rare case is the so-called **Liouvillian systems**. These are systems for which differential flatness properties hold for part of their state vector (constituting a flat subsystem) while the non-flat state variables can be obtained by integration of the elements of the 5 flat subsystem.







#### 3.1. Transformation of MIMO nonlinear systems into the Brunovsky form

The **initial MIMO nonlinear system** is taken to be in the generic form:

$$\dot{x} = f(x, u)$$

It is assumed now that after defining the flat outputs of the initial MIMO nonlinear system, and after expressing the system state variables and control inputs as functions of the flat output and of the associated derivatives, the system can be transformed in the Brunovsky canonical form

$$\begin{aligned} \dot{x}_1 &= x_2 \\ \dot{x}_2 &= x_3 \\ & \cdots \\ \dot{x}_{n_1-1} &= x_{n_1} \\ \dot{x}_{n_1} &= f_1(x) + \sum_{j=1}^p g_{1_j}(x) u_j + d_1 \\ & \vdots \\ \dot{x}_{n_1+1} &= x_{n_1+2} \\ \dot{x}_{n_1+2} &= x_{n_1+3} \\ & \cdots \\ \dot{x}_{p-1} &= x_p \\ \dot{x}_p &= f_p(x) + \sum_{j=1}^p g_{p_j}(x) u_j + d_p \end{aligned}$$

$$= [x_1, \cdots, x_n]^T \quad : \text{ is the state vector}$$

$$= [v_1, \cdots, v_p]^T \quad : \text{ is the inputs vector}$$

$$= [y_1, \cdots, y_p]^T \quad : \text{ is the inputs vector}$$

y x

 $\mathcal{U}$  :

 $\mathcal{Y}$ 





#### 3.1. Transformation of MIMO nonlinear systems into the Brunovsky form

Next **the following vectors and matrices** can be defined

$$f(x) = [f_1(x), ..., f_n(x)]^T$$

$$g(x) = [g_1(x), ..., g_n(x)]^T$$
with  $g_i(x) = [g_{1i}(x), ..., g_{pi}(x)]^T$ 

$$A = diag[A_1, ..., A_p], B = diag[B_1, ..., B_p]$$

$$C^T = diag[C_1, ..., C_p], d = [d_1, ..., d_p]^T$$

where matrix A has the **MIMO canonical form**, i.e. with elements

$$A_{i} = \begin{bmatrix} 0 & 1 & 0 & \dots & 0 \\ 0 & 0 & 1 & \dots & 0 \\ \dots & \dots & \dots & \dots & \dots \\ 0 & 0 & 0 & \dots & 1 \\ 0 & 0 & 0 & \dots & 0 \end{bmatrix}_{r_{i} \times r_{i}}$$
$$= \begin{bmatrix} 0 & 0 & \dots & 0 & 1 \end{bmatrix}_{1 \times r_{i}} \qquad C_{i} = \begin{bmatrix} 1 & 0 & \dots & 0 & 0 \end{bmatrix}_{1 \times r_{i}}$$

 $B_i^T$ 

Thus, the initial nonlinear system can be written in the **state-space form** 

$$\hat{x} = Ax + B[f(x) + g(x)u + \hat{d}]$$
  
$$y = Cx$$



or equivalently in the state space form

 $\dot{x} = Ax + Bv + B\vec{d}$ y = Cx



where v = f(x) + g(x)u

For the generic case of the **MIMO nonlinear system** it is assumed that the functions f(x) and g(x) are unknown and have to be approximated by **nonlinear regressors** (e.g. neuro-fuzzy networks)

The **reference setpoints** for the system's outputs

#### 3.1. Transformation of MIMO nonlinear systems into the Brunovsky form

Thus, the nonlinear system can be written in state-space form

$$\dot{x} = Ax + B[f(x) + g(x)u + \bar{d}]$$
$$y = C^T x$$

which equivalently  $\dot{w} = Ax + Bv + B\bar{d}$ can be written as  $y = C^T x$ 

 $\dot{v} = Ax + Bv + B \overline{d}$  where v = f(x) + g(x)u.  $y = C^T x$ 

 $y_1, \cdots, y_p \in$ 

are denoted as 3/1m, ..., 3/pm and the associated tracking errors are defined as

 $egin{aligned} e_1 &= y_1 - y_{1m} \ e_2 &= y_2 - y_{2m} \ & \dots \ e_p &= y_p - y_{pm} \end{aligned}$ 



The error vector of the outputs of the transformed MIMO system is denoted as

$$E_1 = [e_1, \cdots, e_p]^T$$
$$y_m = [y_{1m}, \cdots, y_{pm}]^T$$
$$\dots$$
$$y_m^{(r)} = [y_{1m}^{(r)}, \cdots, y_{pm}^{(r)}]^T$$



#### 3.2. Control law under measurable state vector

The control signal v = f(x) + g(x)u of the MIMO nonlinear system contains the unknown nonlinear functions f(x) and g(x) which can be approximated by

$$\hat{f}(x|\theta_f) = \Phi_f(x)\theta_f, \quad \hat{g}(x|\theta_g) = \Phi_g(x)\theta_g$$

where

$$\mathbf{e} \quad \Phi_f(\mathbf{x}) = \left(\xi_f^1(\mathbf{x}), \xi_f^2(\mathbf{x}), \cdots, \xi_f^n(\mathbf{x})\right)^T,$$
  
$$\xi_f^{i}(\mathbf{x}) = \left(\phi_f^{i,1}(\mathbf{x}), \phi_f^{i,2}(\mathbf{x}), \cdots, \phi_f^{i,N}(\mathbf{x})\right)$$

thus giving

$$\Phi_f(x) = \begin{pmatrix} \phi_f^{1,1}(x) & \phi_f^{1,2}(x) & \cdots & \phi_f^{1,N}(x) \\ \phi_f^{2,1}(x) & \phi_f^{2,2}(x) & \cdots & \phi_f^{2,N}(x) \\ \cdots & \cdots & \cdots \\ \phi_f^{n,1}(x) & \phi_f^{n,2}(x) & \cdots & \phi_f^{n,N}(x) \end{pmatrix}$$

while the weights vector is defined as  $\theta_f^T = (\theta_f^1, \theta_f^2, \cdots , \theta_f^N)$ 





#### 3. State-space modelling of MIMO nonlinear systems

#### 3.2. Control law under measurable state vector

Similarly, it holds  $\Phi_{\mathcal{E}}(x) = \left(\xi_{\mathcal{E}}^{1}(x), \xi_{\mathcal{E}}^{2}(x), \cdots, \xi_{\mathcal{E}}^{N}(x)\right)^{T}$ 

$$\xi_{g}^{i}(x) = (\phi_{g}^{i,1}(x), \phi_{g}^{i,2}(x), \cdots, \phi_{g}^{i,N}(x)).$$

thus giving

$$\Phi_{g}(x) = \begin{pmatrix} \phi_{g}^{1,1}(x) & \phi_{g}^{1,2}(x) & \cdots & \phi_{g}^{1,N}(x) \\ \phi_{g}^{2,1}(x) & \phi_{g}^{2,2}(x) & \cdots & \phi_{g}^{2,N}(x) \\ \cdots & \cdots & \cdots & \cdots \\ \phi_{g}^{n,1}(x) & \phi_{g}^{n,2}(x) & \cdots & \phi_{g}^{n,N}(x) \end{pmatrix}$$



while the weights vector is defined as  $\theta_g = \left(\theta_g^1, \theta_g^2, \cdots, \theta_g^p\right)^T$ .

However, here each row of  $\theta_{\rm E}$  is vector thus giving

$$\theta_g = \begin{pmatrix} \theta_{g_1}^1 & \theta_{g_1}^2 & \dots & \theta_{g_1}^p \\ \theta_{g_2}^1 & \theta_{g_2}^2 & \dots & \theta_{g_2}^p \\ \dots & \dots & \dots & \dots \\ \theta_{g_N}^1 & \theta_{g_N}^2 & \dots & \theta_{g_N}^p \end{pmatrix}.$$



If the state variables of the system are available for measurement then a **state-feedback** control law can be formulated as

$$u = \hat{g}^{-1}(x|\theta_g) \left[ -\hat{f}(x|\theta_f) + y_m^{(r)} + K_c^T e + u_c \right]$$
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#### 3.2. Control law under non-measurable state vector

The control of the system  $\dot{x} = f(x, u)$  becomes more complicated when the state vector x is not directly measurable and has to be reconstructed through a state observer. The following definitions are used

 $e = x - x_m$  is the error of the state vector

 $\hat{e} = \hat{x} - x_m$  is the error of the estimated state vector

 $\tilde{e} = e - \hat{e} = (x - x_m) - (\hat{x} - x_m)$  is the observation error

When an **observer is used to reconstruct the state vector**, the control law

$$u = \hat{g}^{-1}(\hat{x}|\theta_g) \left[-\hat{f}(\hat{x}|\theta_f) + y_m^{(r)} - K^T \hat{e} + u_c\right]$$

By applying the previous feedback control law one obtains the closed-loop dynamics

 $\begin{aligned} y^{(r)} &= f(x) + g(x)\hat{g}^{-1}(\hat{x})[-\hat{f}(\hat{x}) + y_m^{(r)} - K^T\hat{e} + u_e] + d \Rightarrow \\ y^{(r)} &= f(x) + [g(x) - \hat{g}(\hat{x}) + \hat{g}(\hat{x})]\hat{g}^{-1}(\hat{x})[-\hat{f}(\hat{x}) + y_m^{(r)} - K^T\hat{e} + u_e] + d \Rightarrow \\ y^{(r)} &= [f(x) - \hat{f}(\hat{x})] + [g(x) - \hat{g}(\hat{x})]u + y_m^{(r)} - K^T\hat{e} + u_e + d \end{aligned}$ 

It holds  $\varepsilon = x - x_m \Rightarrow y^{(r)} = \varepsilon^{(r)} + y_m^{(r)}$ 

and by substituting  $\frac{1}{3}$  in the **previous feedback control loop dynamics** gives





#### 3. State-space modelling of MIMO nonlinear systems

#### 3.2. Control law under non-measurable state vector

the tracking error dynamics

$$\begin{split} e^{(r)} + y_m^{(r)} &= y_m^{(r)} - K^T \hat{e} + u_e + [f(x) - \hat{f}(\hat{x})] + \\ &+ [g(x) - \hat{g}(\hat{x})]u + d \end{split}$$

 $\dot{e} = Ae - BK^T \hat{e} + Bu_e + B\{[f(x) - \hat{f}(\hat{x})] +$ 

 $+[q(x) - \hat{q}(\hat{x})]u + d\}$ 

 $\varepsilon_1 = C^T \varepsilon$ 

or equivalently





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where  $e = [e^1, e^2, \cdots, e^p]^T$  with  $e^i = [e_i, \dot{e}_i, \ddot{e}_i, \cdots, e_i^{*i-1}]^T, i = 1, 2, \cdots, p$ 

and equivalently  $\hat{e} = [\hat{e}^1, \hat{e}^2, \cdots, \hat{e}^p]^T$  with  $\hat{e}^i = [\hat{e}_i, \hat{e}_i, \hat{e}_i, \hat{e}_i, \cdots, \hat{e}_i^{*i-1}]^T$ ,  $i = 1, 2, \cdots, p$ .

A state observer is designed as:

$$\dot{\hat{\varepsilon}} = A\hat{\varepsilon} - BK^T\hat{\varepsilon} + K_o[\varepsilon_1 - C^T\hat{\varepsilon}]$$

$$\hat{\varepsilon}_1 = C^T\hat{\varepsilon}$$

$$(B)$$

Α

#### 4.1. Dynamics of the tracking error

Without loss of generality consider a two-input MIMO system:

By applying differential flatness theory, and in the presence of disturbances, the dynamic model of the system comes to the form

> $\ddot{x}_1 = f_1(x,t) + g_1(x,t)u + d_1$  $\ddot{x}_3 = f_2(x,t) + q_2(x,t)u + d_2$

The following **control input** is defined:

$$u = \begin{pmatrix} \hat{g}_1(x,t) \\ \hat{g}_2(x,t) \end{pmatrix}^{-1} \left\{ \begin{pmatrix} \ddot{x}_1^d \\ \ddot{x}_3^d \end{pmatrix} - \begin{pmatrix} \hat{f}_1(x,t) \\ \hat{f}_2(x,t) \end{pmatrix} - \begin{pmatrix} K_1^T \\ K_2^T \end{pmatrix} \varepsilon + \begin{pmatrix} u_{e_1} \\ u_{e_2} \end{pmatrix} \right\} \qquad (D)$$

where:  $[u_{c_1} u_{c_2}]^T$  is a **robust control term** that is used for the compensation of the model's uncertainties as well as of the external disturbances

and:  $\mathcal{K}_{i}^{T} = [k_{1}^{i}, k_{2}^{i}, \cdots, k_{n-1}^{i}, k_{n}^{i}]$  is the feedback gain

Substituting the control input (D) into the system (C)

one obtains





#### 4.1. Dynamics of the tracking error

Moreover, using again Eq. (D) one obtains the **tracking error dynamics** 

 $\begin{pmatrix} \ddot{e}_1 \\ \ddot{e}_3 \end{pmatrix} = \begin{pmatrix} f_1(x,t) - f_1(x,t) \\ f_2(x,t) - f_2(x,t) \end{pmatrix} + \begin{pmatrix} g_1(x,t) - \hat{g}_1(x,t) \\ g_2(x,t) - \hat{g}_2(x,t) \end{pmatrix} u - \begin{pmatrix} K_1^T \\ K_2^T \end{pmatrix} e + \begin{pmatrix} u_{c_1} \\ u_{c_2} \end{pmatrix} + \begin{pmatrix} d_1 \\ d_2 \end{pmatrix}$ 

The **approximation error** is defined as:

$$w = \begin{pmatrix} f_1(x,t) - \hat{f}_1(x,t) \\ f_2(x,t) - \hat{f}_2(x,t) \end{pmatrix} + \begin{pmatrix} g_1(x,t) - \hat{g}_1(x,t) \\ g_2(x,t) - \hat{g}_2(x,t) \end{pmatrix} u$$



Using matrices A,B,K, and considering that **the estimated state vector is used in the control loop** the following description of the tracking error dynamics is obtained:

$$\dot{e} = Ae - BK^{T}\hat{e} + Bu_{e} + B\left\{ \begin{pmatrix} f_{1}(x,t) - \hat{f}_{1}(\hat{x},t) \\ f_{2}(x,t) - \hat{f}_{2}(\hat{x},t) \end{pmatrix} + \begin{pmatrix} g_{1}(x,t) - \hat{g}_{1}(\hat{x},t) \\ g_{2}(x,t) - \hat{g}_{2}(\hat{x},t) \end{pmatrix} u + \tilde{d} \right\}$$

When the estimated state vector is used in the loop the approximation error is written as

$$w = \begin{pmatrix} f_1(x,t) - \hat{f}_1(\hat{x},t) \\ f_2(x,t) - \hat{f}_2(\hat{x},t) \end{pmatrix} + \begin{pmatrix} g_1(x,t) - \hat{g}_1(\hat{x},t) \\ g_2(x,t) - \hat{g}_2(\hat{x},t) \end{pmatrix} u$$

while the tracking error dynamics becomes

$$\dot{e} = Ae - BK^T \hat{e} + Bu_c + Bw + B\hat{d}$$



#### 4.2. Dynamics of the observation error

The observation error is defined as:  $\bar{\varepsilon} = \varepsilon - \hat{\varepsilon} = \omega - \hat{\omega}$ .

By subtracting Eq. B from Eq. A one obtains:



$$\dot{\varepsilon} - \dot{\hat{\varepsilon}} = A(\varepsilon - \hat{\varepsilon}) + Bu_{\varepsilon} + B\{[f(x, t) - \hat{f}(\hat{x}, t)] + [g(x, t) - \hat{g}(\hat{x}, t)]u + \bar{d}\} - K_o C^T (\varepsilon - \hat{\varepsilon})$$

 $\varepsilon_1 - \hat{\varepsilon}_1 = C^T (\varepsilon - \hat{\varepsilon})$ 

or equivalently:

$$\dot{ar{e}} = Aar{e} + Bu_e + B\{[f(x,t) - \hat{f}(\hat{x},t)] + [g(x,t) - \hat{g}(\hat{x},t)]u + ar{d}\} - K_o C^Tar{e}$$
 $ar{e}_1 = C^Tar{e}$ 

which can be also written as:

$$\begin{split} & \dot{\bar{e}} = (A - K_o C^T) \bar{e} + B u_e + B w + \bar{d} \\ & \bar{e}_1 = C^T \bar{e} \end{split}$$



#### 4.3. Approximation of the unknown system dynamics

Next, the first of the approximators of the unknown system dynamics is defined

$$\hat{f}(\hat{x}) = \begin{pmatrix} \hat{f}_1(\hat{x}|\theta_f) \ \hat{x} \in R^{4 \times 1} \ \hat{f}_1(\hat{x}|\theta_f) \ \in \ R^{1 \times 1} \\ \hat{f}_2(\hat{x}|\theta_f) \ \hat{x} \in R^{4 \times 1} \ \hat{f}_2(\hat{x}|\theta_f) \ \in \ R^{1 \times 1} \end{pmatrix}$$





containing kernel functions  $\phi_f^{i,j}(\hat{x}) = \frac{\prod_{j=1}^n \mu_{A_j}^i(\hat{x}_j)}{\sum_{i=1}^N \prod_{j=1}^n \mu_{A_j}^i(\hat{x}_j)}$ 

where  $\mu_{A_{\vec{2}}}(\hat{x})$  are fuzzy membership functions appearing in the antecedent part of the *I-th* fuzzy rule

#### 4.3. Approximation of the unknown system dynamics

Similarly, the second of the approximators of the unknown system dynamics is defined

$$\hat{g}(\hat{x}) = \begin{pmatrix} \hat{g}_{1}(\hat{x}|\theta_{g}) \ \hat{x} \in R^{4 \times 1} \ \hat{g}_{1}(\hat{x}|\theta_{g}) \ \in \ R^{1 \times 2} \\ \hat{g}_{2}(\hat{x}|\theta_{g}) \ \hat{x} \in R^{4 \times 1} \ \hat{g}_{2}(\hat{x}|\theta_{g}) \ \in \ R^{1 \times 2} \end{pmatrix}$$

The values of the weights that result in optimal approximation are

$$\begin{split} \theta_f^* &= \arg \ \min_{\theta_f \in \mathcal{M}_{\theta_f}} [ \sup_{\vartheta \in U_2} (f(x) - \hat{f}(\hat{x}|\theta_f)) \\ \theta_g^* &= \arg \ \min_{\theta_g \in \mathcal{M}_{\theta_g}} [ \sup_{\vartheta \in U_2} (g(x) - \hat{g}(\hat{x}|\theta_g)) ] \end{split}$$

The variation ranges for the weights are given by

$$\begin{split} &M_{\boldsymbol{\theta}_f} = \{\boldsymbol{\theta}_f \!\in\! \! \boldsymbol{R}^h: \; ||\boldsymbol{\theta}_f|| \!\leq\! m_{\boldsymbol{\theta}_f} \} \\ &M_{\boldsymbol{\theta}_g} = \{\boldsymbol{\theta}_g \!\in\! \! \boldsymbol{R}^h: \; ||\boldsymbol{\theta}_g|| \!\leq\! m_{\boldsymbol{\theta}_g} \} \end{split}$$

The **value of the approximation error** that corresponds to the optimal values of the weights vectors is

$$w = \left(f(x,t) - \hat{f}(\hat{x}|\theta_f^*)\right) + \left(g(x,t) - \hat{g}(\hat{x}|\theta_g^*)\right)u$$





#### 4.3. Approximation of the unknown system dynamics

which is next written as

$$w = \left(f(x,t) - \hat{f}(\hat{x}|\theta_f) + \hat{f}(\hat{x}|\theta_f) - \hat{f}(\hat{x}|\theta_f^*)\right) + \left(g(x,t) - \hat{g}(\hat{x}|\theta_g) + \hat{g}(\hat{x}|\theta_g) - \hat{g}(\hat{x}|\theta_g^*)\right)u$$

which can be also written in the following form

with

$$w_a = \{[f(x,t) - \hat{f}(\hat{x}|\boldsymbol{\theta}_f)] + [g(x,t) - \hat{g}(\hat{x}|\boldsymbol{\theta}_g)]\}u$$

and

$$w_b = \{ [\hat{f}(\hat{x}|\theta_f) - \hat{f}(\hat{x}|\theta_f^*)] + [\hat{g}(\hat{x},\theta_g) - \hat{g}(\hat{x}|\theta_g^*)] \} u$$

Moreover, the following weights error vectors are defined

 $w = (w_a + w_b)$ 

$$\begin{split} \bar{\boldsymbol{\theta}}_f &= \boldsymbol{\theta}_f - \boldsymbol{\theta}_f^* \\ \bar{\boldsymbol{\theta}}_g &= \boldsymbol{\theta}_g - \boldsymbol{\theta}_g^* \end{split}$$



and these denote the distance of the **weights vectors** from the values that provide optimal model estimation

It will be shown that these weights are updated through a gradient method

#### 5. Convergence proof for the optimization method

The following Lyapunov (energy) function is considered:

$$V = \frac{1}{2}\hat{\varepsilon}^T P_1 \hat{\varepsilon} + \frac{1}{2}\bar{\varepsilon}^T P_2 \bar{\varepsilon} + \frac{1}{2\gamma_1}\bar{\theta}_f^T \bar{\theta}_f + \frac{1}{2\gamma_2}tr[\bar{\theta}_g^T \bar{\theta}_g]$$

The selection of the **Lyapunov function** is based on the following principle of indirect adaptive control

 $\hat{\varepsilon} : \lim_{t \to \infty} \hat{x}(t) = x_d(t)$  this results  $\bar{\varepsilon} : \lim_{t \to \infty} \hat{x}(t) = x(t)$ .

By deriving the Lyapunov function with respect to time one obtains:

$$\begin{split} \dot{V} &= \frac{1}{2}\dot{\hat{e}}^T P_1 \hat{e} + \frac{1}{2}\dot{\hat{e}}^T P_1 \dot{\hat{e}} + \frac{1}{2}\dot{\bar{e}}^T P_2 \bar{e} + \frac{1}{2}\bar{e}^T P_2 \dot{\bar{e}} + \\ &+ \frac{1}{\gamma_4} \dot{\bar{\theta}}_f^T \bar{\theta}_f + \frac{1}{\gamma_5} tr[\dot{\bar{\theta}}_g^T \bar{\theta}_g] \Rightarrow \end{split}$$



 $\lim_{t \to \infty} x(t) = x_d(t)$ 

$$\begin{split} \dot{V} &= \frac{1}{2} \{ (A - BK^{T}) \hat{e} + K_{o}C^{T}\bar{e} \}^{T} P_{1} \hat{e} + \frac{1}{2} \hat{e}^{T} P_{1} \{ (A - BK^{T}) \hat{e} + K_{o}C^{T}\bar{e} \} + \\ &+ \frac{1}{2} \{ (A - K_{o}C^{T})\bar{e} + Bu_{e} + B\bar{d} + Bw \}^{T} P_{2}\bar{e} + \\ &+ \frac{1}{2} \bar{e}^{T} P_{2} \{ (A - K_{o}C^{T})\bar{e} + Bu_{e} + B\bar{d} + Bw \} + \\ &+ \frac{1}{\gamma_{1}} \dot{\bar{\theta}}_{f}^{T} \bar{\theta}_{f} + \frac{1}{\gamma_{2}} tr[\dot{\bar{\theta}}_{g}^{T} \bar{\theta}_{g}] \Rightarrow \end{split}$$



The previous equation is rewritten as:

$$\begin{split} \dot{V} &= \frac{1}{2} \{ \hat{\varepsilon}^T (A - BK^T)^T + \bar{\varepsilon}^T CK_o^T \} P_1 \hat{\varepsilon} + \frac{1}{2} \hat{\varepsilon}^T P_1 \{ (A - BK^T) \hat{\varepsilon} + K_o C^T \bar{\varepsilon} \} + \\ &+ \frac{1}{2} \{ \bar{\varepsilon}^T (A - K_o C^T)^T + u_e^T B^T + w^T B^T + \bar{d}^T B^T \} P_2 \bar{\varepsilon} + \\ &\frac{1}{2} \bar{\varepsilon}^T P_2 \{ (A - K_o C^T) \bar{\varepsilon} + Bu_e + Bw + B\bar{d} \} + \frac{1}{\gamma_1} \dot{\bar{\theta}}_f^T \bar{\theta}_f + \frac{1}{\gamma_2} tr[\dot{\bar{\theta}}_g^T \bar{\theta}_g] \Rightarrow \\ &\text{ which finally takes the form:} \end{split}$$

$$\begin{split} \dot{V} &= \frac{1}{2} \hat{\varepsilon}^{T} (A - BK^{T})^{T} P_{1} \hat{\varepsilon} + \frac{1}{2} \bar{\varepsilon}^{T} CK_{\circ}^{T} P_{1} \hat{\varepsilon} + \\ &+ \frac{1}{2} \hat{\varepsilon}^{T} P_{1} (A - BK^{T}) \hat{\varepsilon} + \frac{1}{2} \hat{\varepsilon}^{T} P_{1} K_{\circ} C^{T} \bar{\varepsilon} + \\ &+ \frac{1}{2} \bar{\varepsilon}^{T} (A - K_{\circ} C^{T})^{T} P_{2} \bar{\varepsilon} + \frac{1}{2} (u_{c}^{T} + w^{T} + \bar{d}^{T}) B^{T} P_{2} \bar{\varepsilon} + \\ &+ \frac{1}{2} \bar{\varepsilon}^{T} P_{2} (A - K_{\circ} C^{T}) \bar{\varepsilon} + \frac{1}{2} \bar{\varepsilon}^{T} P_{2} B (u_{c} + w + \bar{d}) + \\ &+ \frac{1}{\gamma_{1}} \dot{\bar{\theta}}_{f}^{T} \bar{\theta}_{f} + \frac{1}{\gamma_{2}} tr [\dot{\bar{\theta}}_{g}^{T} \bar{\theta}_{g}] \end{split}$$





**Assumption 1:** For given positive definite matrices Q1 and Q2 there exist positive definite matrices P1 and P2, which are the solution of the following **Riccati equations** 

$$(A - BK^{T})^{T}P_{1} + P_{1}(A - BK^{T}) + Q_{1} = 0$$
$$(A - K_{o}C^{T})^{T}P_{2} + P_{2}(A - K_{o}C^{T}) - P_{2}B(\frac{2}{r} - \frac{1}{o^{2}})B^{T}P_{2} + Q_{2} = 0$$

$$\begin{split} \dot{V} &= \frac{1}{2} \hat{e}^{T} \{ (A - BK^{T})^{T} P_{1} + P_{1} (A - BK^{T}) \} \hat{e} + \bar{e}^{T} CK_{o}^{T} P_{1} \hat{e} + \\ &+ \frac{1}{2} \bar{e}^{T} \{ (A - K_{o}C^{T})^{T} P_{2} + P_{2} (A - K_{o}C^{T}) \} \bar{e} + \\ &+ \bar{e}^{T} P_{2} B(u_{o} + w + \bar{d}) + \frac{1}{\gamma_{1}} \dot{\theta}_{f}^{T} \bar{\theta}_{f} + \frac{1}{\gamma_{2}} tr[\dot{\theta}_{g}^{T} \bar{\theta}_{g}] \end{split}$$

By substituting the relations described by the previous **Riccati equations** into the derivative

or:  

$$\dot{V} = -\frac{1}{2}\hat{e}^{T}Q_{1}\hat{e} + \bar{e}^{T}CK_{o}^{T}P_{1}\hat{e} - \frac{1}{2}\bar{e}^{T}\{Q_{2} - P_{2}B(\frac{2}{n} - \frac{1}{\rho^{2}})B^{T}P_{2}\}\bar{e} + \bar{e}^{T}P_{2}B(u_{o} + w + \bar{d}) + \frac{1}{\gamma_{1}}\dot{\theta}_{f}^{T}\bar{\theta}_{f} + \frac{1}{\gamma_{2}}tr[\dot{\bar{\theta}}_{g}^{T}\bar{\theta}_{g}]$$

The supervisory control term  $u_c$  consists of two terms  $u_a$  and  $u_b$ 

The first term % is

$$u_a = -\frac{1}{r}\tilde{e}^T P_2 B + \Delta u_a$$

where assuming that the measurable elements of vector  $\tilde{e}$  are





$$\{ ilde{e}_1, ilde{e}_3,\cdots, ilde{e}_k\}$$

#### 5. Convergence proof for the optimization method

of the Lyapunov function one gets:

#### 5. Convergence proof for the optimization method

The term  $\Delta u_a$  is such that  $-\frac{1}{r}\tilde{e}^T P_2 B + \Delta u_a = -\frac{1}{r} \begin{pmatrix} p_{11}\tilde{e}_1 + p_{13}\tilde{e}_3 + \dots + p_{1k}\tilde{e}_k \\ p_{13}\tilde{e}_1 + p_{33}\tilde{e}_3 + \dots + p_{3k}\tilde{e}_k \\ \dots \dots \dots \\ p_{1k}\tilde{e}_1 + p_{3k}\tilde{e}_3 + \dots + p_{kk}\tilde{e}_k \end{pmatrix}$ 



 $u_{\alpha}$  is an  $H_{\infty}$  control used for the **compensation of the approximation error** w and the additive disturbance  $\overline{d}$  (the control term  $u_{\alpha}$  has been chosen so as to satisfy the condition

The previous relation finally stands for a product between the measurable state vector elements  $\{\tilde{e}_1, \tilde{e_3}, \dots, \tilde{e_k}\}$  and the elements of matrix  $P_2$  which is obtained from the solution of the previous Riccati equation.

The control term  $u_b$  is given by



$$u_b = -\left[(P_2B)^T(P_2B)\right]^{-1}(P_2B)^TCK_o^TP_1\hat{\varepsilon}$$

 $w_{0}$  is a control used for the **compensation of the observation error** (the control term  $w_{0}$  has been chosen so as to satisfy the condition  $\tilde{e}^{T}P_{2}Bw_{0} = -\tilde{e}^{T}CK_{o}^{T}P_{1}\hat{e}$ .

#### 5. Convergence proof for the optimization method

The **optimization-based control scheme** is depicted in the following diagram







By substituting the supervisory control term in the **derivative of the Lyapunov function** one obtains

$$\begin{split} \dot{V} &= -\frac{1}{2} \dot{\varepsilon}^T Q_1 \dot{\varepsilon} + \bar{\varepsilon}^T C K_o^T P_1 \dot{\varepsilon} - \frac{1}{2} \bar{\varepsilon}^T Q_2 \bar{\varepsilon} + \frac{1}{r} \bar{\varepsilon}^T P_2 B B^T P_2 \bar{\varepsilon} - \frac{1}{2\rho^2} \bar{\varepsilon}^T P_2 B B^T P_2 \bar{\varepsilon} + \\ &+ \bar{\varepsilon}^T P_2 B u_a + \bar{\varepsilon}^T P_2 B u_b + \bar{\varepsilon}^T P_2 B (w + \bar{d}) + \frac{1}{\gamma_1} \dot{\beta}_f^T \bar{\theta}_f + \frac{1}{\gamma_2} tr[\dot{\bar{\theta}}_g^T \bar{\theta}_g] \end{split}$$

# or equivalently $$\begin{split} \dot{V} &= -\frac{1}{2}\hat{e}^{T}Q_{1}\hat{e} - \frac{1}{2}\tilde{e}^{T}Q_{2}\tilde{e} - \frac{1}{2\rho^{2}}\tilde{e}^{T}P_{2}BB^{T}P_{2}\tilde{e} + \\ &+ \tilde{e}^{T}P_{2}B(w + \tilde{d} + \Delta u_{a}) + \frac{1}{\gamma_{1}}\dot{\tilde{\theta}}_{f}^{T}\tilde{\theta}_{f} + \frac{1}{\gamma_{2}}tr[\dot{\tilde{\theta}}_{g}^{T}\tilde{\theta}_{g}] \end{split}$$

Besides, about the **adaptation of the weights** of the neurofuzzy approximator it holds

$$\dot{\theta}_f = \dot{\theta}_f - \dot{\theta}_f^* = \dot{\theta}_f \qquad \qquad \dot{\theta}_g = \dot{\theta}_g - \dot{\theta}_g^* = \dot{\theta}_g.$$



A gradient-based update is applied to the approximator's weights

$$\dot{\theta}_f = -\gamma_1 \Phi(\hat{x})^T B^T P_2 \bar{e} \\ \dot{\theta}_g = -\gamma_2 \Phi(\hat{x})^T B^T P_2 \bar{e} u^T$$

### Gradient-based optimization

The gradient update scheme is defined in a manner that assures that the first derivative of the Lyapunov function will remain negative, and thus the Lyapunov function will be monotonously decreasing.

By substituting the above relations in the derivative of the Lyapunov function one obtains

$$\begin{split} \dot{V} &= -\frac{1}{2}\hat{e}^{T}Q_{1}\hat{e} - \frac{1}{2}\tilde{e}^{T}Q_{2}\tilde{e} - \frac{1}{2\rho^{2}}\tilde{e}^{T}P_{2}BB^{T}P_{2}\tilde{e} + \\ B^{T}P_{2}\tilde{e}(w + \tilde{d} + \Delta u_{a}) + \frac{1}{\gamma_{1}}(-\gamma_{1})\tilde{e}^{T}P_{2}B\Phi(\hat{x})(\theta_{f} - \theta_{f}^{*}) + \\ \frac{1}{\gamma_{2}}(-\gamma_{2})tr[u\tilde{e}^{T}P_{2}B(\hat{g}(\hat{x}|\theta_{g}) - \hat{g}(\hat{x}|\theta_{g}^{*})] \end{split}$$



To continue with the **convergence proof for the proposed optimization method** it is taken into account that

$$u \in R^{2 imes 1}$$
 and  $ar{e}^T PB(\hat{g}(w| heta_g) - \hat{g}(w| heta_g^*)) \in R^{1 imes 2}$ 

one gets

$$\dot{V} = -\frac{1}{2}\hat{e}^{T}Q_{1}\hat{e} - \frac{1}{2}\tilde{e}^{T}Q_{2}\tilde{e} - \frac{1}{2\rho^{2}}\tilde{e}^{T}P_{2}BB^{T}P_{2}\tilde{e} + B^{T}P_{2}\tilde{e}(w + \tilde{d} + \Delta u_{a}) + \frac{1}{\gamma_{1}}(-\gamma_{1})\tilde{e}^{T}P_{2}B\Phi(\hat{x})(\theta_{f} - \theta_{f}^{*}) + \frac{1}{\gamma_{2}}(-\gamma_{2})tr[\tilde{e}^{T}P_{2}B(\hat{g}(\hat{x}|\theta_{g}) - \hat{g}(\hat{x}|\theta_{g}^{*}))u]$$

Since

it holds

$$e^{-} F_2 \mathcal{B}(g(x|\sigma_g) - g(x|\sigma_g)) u \in R^{-}$$

TT TO D(A(A)A) A(A)A\*)) = T1X1

$$\begin{aligned} &\tilde{e}^T P_2 B(\hat{g}(x|\theta_g) - \hat{g}(x|\theta_g^*)u) = \\ &= \bar{e}^T P_2 B(\hat{g}(x|\theta_g) - \hat{g}(x|\theta_g^*))u \end{aligned}$$

Therefore, one finally obtains

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$$\begin{split} \dot{V} &= -\frac{1}{2} \hat{e}^{T} Q_{1} \hat{e} - \frac{1}{2} \tilde{e}^{T} Q_{2} \tilde{e} - \frac{1}{2\rho^{2}} \tilde{e}^{T} P_{2} B B^{T} P_{2} \tilde{e} + \\ B^{T} P_{2} \tilde{e}(w + \tilde{d} + \Delta u_{a}) + \frac{1}{\gamma_{1}} (-\gamma_{1}) \tilde{e}^{T} P_{2} B \Phi(\hat{x}) (\theta_{f} - \theta_{f}^{*}) + \\ \frac{1}{\gamma_{2}} (-\gamma_{2}) \tilde{e}^{T} P_{2} B(\hat{g}(\hat{x}|\theta_{g}) - \hat{g}(\hat{x}|\theta_{g}^{*})) u \end{split}$$

Next, the following approximation error is defined

$$w_{\alpha} = [\hat{f}(\hat{x}|\theta_f^*) - \hat{f}(\hat{x}|\theta_f)] + [\hat{g}(\hat{x}|\theta_g^*) - \hat{g}(\hat{x}|\theta_g)]u$$





Thus, one obtains

$$\begin{split} \dot{V} &= -\frac{1}{2} \hat{e}^T Q_1 \hat{e} - \frac{1}{2} \tilde{e}^T Q_2 \tilde{e} - \frac{1}{2\rho^2} \tilde{e}^T P_2 B B^T P_2 \tilde{e} + \\ &+ B^T P_2 \tilde{e} (w + \tilde{d} + \Delta u_a) + \tilde{e}^T P_2 B w_\alpha \end{split}$$

Denoting the **aggregate approximation error** and disturbances vector as

$$w_1 = w + \tilde{d} + w_\alpha + \Delta u_a$$

the derivative of the Lyapunov function becomes

$$\dot{V} = -\frac{1}{2}\hat{\varepsilon}^T Q_1 \hat{\varepsilon} - \frac{1}{2}\bar{\varepsilon}^T Q_2 \bar{\varepsilon} - \frac{1}{2\rho^2}\bar{\varepsilon}^T P_2 B B^T P_2 \bar{\varepsilon} + \bar{\varepsilon}^T P_2 B w_1$$

which in turn is written as

$$\begin{split} \dot{V} &= -\frac{1}{2} \hat{e}^T Q_1 \hat{e} - \frac{1}{2} \bar{e}^T Q_2 \bar{e} - \frac{1}{2\rho^2} \bar{e}^T P_2 B B^T P_2 \bar{e} + \\ &+ \frac{1}{2} \bar{e}^T P B w_1 + \frac{1}{2} w_1^T B^T P_2 \bar{e} \end{split}$$

Lemma: The following inequality holds

$$\begin{array}{l} \frac{1}{2}\bar{e}^T P_2 B w_1 + \frac{1}{2} w_1^T B^T P_2 \bar{e} - \frac{1}{2\rho^2} \bar{e}^T P_2 B B^T P_2 \bar{e} \\ \leq \frac{1}{2} \rho^2 w_1^T w_1 \end{array}$$







#### **Proof:**

The binomial  $(\rho a - \frac{1}{\rho}b)^2 \ge 0$  is considered. Expanding the left part of the above inequality one gets

$$\begin{split} \rho^2 a^2 &+ \frac{1}{\rho^2} b^2 - 2ab \ge 0 \Rightarrow \\ \frac{1}{2} \rho^2 a^2 &+ \frac{1}{2\rho^2} b^2 - ab \ge 0 \Rightarrow \\ ab &- \frac{1}{2\rho^2} b^2 \le \frac{1}{2} \rho^2 a^2 \Rightarrow \\ \frac{1}{2} ab &+ \frac{1}{2} ab - \frac{1}{2\rho^2} b^2 \le \frac{1}{2} \rho^2 a^2 \end{split}$$



By substituting  $a = w_1$  and  $b = \tilde{e}^T P_2 B$  one gets

$$\frac{\frac{1}{2}w_{1}^{T}B^{T}P_{2}\bar{e} + \frac{1}{2}\bar{e}^{T}P_{2}Bw_{1} - \frac{1}{2\rho^{2}}\bar{e}^{T}P_{2}BB^{T}P_{2}\bar{e}}{\leq \frac{1}{2}\rho^{2}w_{1}^{T}w_{1}}$$

Moreover, by substituting the above inequality into the **derivative of the Lyapunov** function one gets

$$\dot{V} \le -\frac{1}{2} \hat{\varepsilon}^T Q_1 \hat{\varepsilon} - \frac{1}{2} \bar{\varepsilon}^T Q_2 \bar{\varepsilon} + \frac{1}{2} \rho^2 w_1^T w_1$$

which is also written as

 $\dot{V} \leq -\frac{1}{2}E^TQE + \frac{1}{2}\rho^2 w_1^T w_1$ 



with

Hence, the  $H_{\infty}$  beformance criterion is derived. For sufficiently small  $\rho$  the inequality will be true and the  $H_{\infty}$  racking criterion will be satisfied. In that case, the integration of 'V from 0 to T gives

$$\begin{split} \int_0^T &\dot{V}(t) dt \leq -\frac{1}{2} \int_0^T ||E||^2 dt + \frac{1}{2} \rho^2 \int_0^T ||w_1||^2 dt \Rightarrow \\ 2V(T) - 2V(0) \leq -\int_0^T ||E||_Q^2 dt + \rho^2 \int_0^T ||w_1||^2 dt \Rightarrow \\ 2V(T) + \int_0^T ||E||_Q^2 dt \leq 2V(0) + \rho^2 \int_0^T ||w_1||^2 dt \end{split}$$



It is assumed that there exists a positive constant  $M_w > 0$  such that

$$\int_0^\infty ||w_1||^2 dt \le M_w$$

Therefore for the integral  $\int_0^T ||E||_Q^2 dt$  one gets

$$\int_0^\infty ||E||_Q^2 dt \le 2V(0) + \rho^2 M_w$$



Thus, the integral  $\int_0^\infty ||E||_Q^2 dt$  is bounded and according to Barbalat's Lemma

$$\lim_{t\to\infty} e(t) = 0$$

and thus global asymptotic stability is also shown for the control loop.

#### 6.1 The model of multi-DOF robotic manipulators

The model of the **robot's dynamics** is a MIMO nonlinear one:



#### 6.1 The model of multi-DOF robotic manipulators



**Defining flat outputs**  $y_1$  and  $y_2$  for which holds

$$y = \begin{bmatrix} \theta_1 & \theta_2 \end{bmatrix} = \begin{bmatrix} x_1 & x_3 \end{bmatrix}$$
  

$$x_1 = \theta_1 \quad x_2 = \dot{\theta}_1 \quad x_3 = \theta_2 \quad x_4 = \dot{\theta}_2$$
  

$$\vdots$$
  

$$x_1 = f_1(x) + g_{11}(x)u_1 + g_{12}(x)u_2$$
  

$$\vdots$$
  

$$x_3 = f_2(x) + g_{21}(x)u_1 + g_{22}(x)u_2$$

$$f_{1}(x) = -N_{11}F_{1}(\theta,\theta) - N_{12}F_{2}(\theta,\theta) - N_{11}G_{1}(\theta) - N_{12}G_{2}(\theta) \in R^{1\times 1}$$

$$g_{1}(x) = \begin{bmatrix} N_{11} & N_{12} \end{bmatrix} \in R^{1\times 2}$$

$$f_{2}(x) = -N_{21}F_{2}(\theta,\dot{\theta}) - N_{22}F_{2}(\theta,\dot{\theta}) - N_{21}G_{1}(\theta) - N_{22}G_{2}(\theta) \in R^{1\times 1}$$

$$g_{1}(x) = \begin{bmatrix} N_{11} & N_{12} \end{bmatrix} \in R^{1\times 2}$$

$$g_2(x) = \begin{bmatrix} N_{21} & N_{22} \end{bmatrix} \in R^{1 \times 2}$$

X





the following **Brunovsky (canonical form) of the robotic system** is finally obtained

$$\begin{bmatrix} \cdot \\ x_1 \\ \cdot \\ x_2 \\ \cdot \\ x_3 \\ \cdot \\ x_4 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 \\ 0 & 0 & 0 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} + \begin{bmatrix} 0 & 0 \\ 1 & 0 \\ 0 & 0 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} v_1 \\ v_2 \end{bmatrix}$$

$$v_1 = f_1(x) + g_{11}(x)u_1 + g_{12}(x)u_2$$
  

$$v_2 = f_2(x) + g_{21}(x)u_1 + g_{22}(x)u_2$$
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#### 6.1 The model of multi-DOF robotic manipulators





The optimization problem has multiple objectives:

- 1) Minimize the modelling error of the system's dynamics
- 2) Minimize the estimation error for the system's state vector
- 3) Minimize the tracking error from the reference setpoints

For the differentially flat MIMO model of the multi-DOF robotic manipulator one gets the equivalent state-space model

$$x_{1} = f_{1}(x,t) + g_{1}(x,t)u + d_{1}$$
  

$$x_{3} = f_{2}(x,t) + g_{2}(x,t)u + d_{2}$$

• For the multi-DOF robot control scheme differential flatness properties hold and one can apply the control scheme analyzed in Sections 3 and 4.

#### 6.2 The model of distributed power generators

The **dynamic model of the distributed power generation units** is assumed to be that of synchronous generators. The modelling approach is also applicable to PMSGs (permanent magnet synchronous generators) which are a special case of synchronous electric machines.

$$\delta = \omega$$
  

$$\dot{\omega} = -\frac{D}{2J}(\omega - \omega_0) + \frac{\omega_0}{2J}(P_m - P_e)$$
(1)

 $P_e$ : active electrical power of the machine S turn angle of the rotor  $P_m$ mechanical power of the machine turn speed of the rotor w synchronous speed damping coefficient D  $\omega_0$ moment of inertia of the rotor  $T_e$ : electromagnetic torque J

The generator's electrical dynamics is:

$$\dot{E}_q' = \frac{1}{T_{d_o}} (E_f - E_q)$$

 $E'_q$  is the quadrature-axis transient voltage (a variable related to the magnetic flux)  $E_q$  is quadrature axis voltage of the generator  $T_{d_o}$  is the direct axis open-circuit transient time constant  $E_f$  is the equivalent voltage in the excitation coil

#### 6.2 The model of distributed power generators

The synchronous generator's model is complemented by a set of algebraic equations:

$$E_q = \frac{x_{d_{\Sigma}}}{x'_{d_{\Sigma}}} E'_q - (x_d - x'_d) \frac{V_s}{x'_{d_{\Sigma}}} \cos(\Delta \delta)$$

$$I_q = \frac{V_s}{x'_{d_{\Sigma}}} \sin(\Delta \delta)$$

$$I_d = \frac{E'_q}{x'_{d_{\Sigma}}} - \frac{V_s}{x'_{d_{\Sigma}}} \cos(\Delta \delta)$$

$$P_e = \frac{V_s E'_q}{x'_{d_{\Sigma}}} \sin(\Delta \delta)$$

$$Q_e = \frac{V_s E'_q}{x'_{d_{\Sigma}}} \cos(\Delta \delta) - \frac{V_s^2}{x_{d_{\Sigma}}}$$

$$V_t = \sqrt{(E'_q - X'_d I_d)^2 + (X'_d I_q)^2}$$





where:  $x_{d_{\Sigma}} = x_{d} + x_{T} + x_{L}$   $x'_{d_{\Sigma}} = x'_{d} + x_{T} + x_{L}$ 

- $x_d$ : direct-axis synchronous reactance
- $x_T$  : reactance of the transformer
- $x'_{d}$  : direct-axis transient reactance
- $x_L$  : transmission line reactance

 $I_d$  and  $I_q$ : direct and quadrature axis currents

- $V_s$  : infinite bus voltage
- $Q_e$ : reactive power of the generator
- $V_t$ : terminal voltage of the generator

#### 6.2 The model of distributed power generators

From Eq. 1 and Eq. 2 one obtains the **dynamic model of the synchronous generator**:

$$\begin{split} \delta &= \omega - \omega_0 \\ \dot{\omega} &= -\frac{D}{2J}(\omega - \omega_0) + \omega_0 \frac{P_m}{2J} - \omega_0 \frac{1}{2J} \frac{V_s E'_q}{x'_{d_{\Sigma}}} sin(\Delta \delta) \\ \dot{E}'_q &= -\frac{1}{T'_d} E'_q + \frac{1}{T_{d_o}} \frac{x_d - x'_d}{x'_{d_{\Sigma}}} V_s cos(\Delta \delta) + \frac{1}{T_{d_o}} E_f \end{split}$$

Moreover, the generator can be written in a state-space form:

$$\dot{x} = f(x) + g(x)u$$

where the state vector is  $x = \begin{pmatrix} \Delta \delta & \Delta \omega & E'_q \end{pmatrix}^T$  and

$$f(x) = \begin{pmatrix} \omega - \omega_0 \\ -\frac{D}{2J}(\omega - \omega_0) + \omega_0 \frac{P_m}{2J} - \omega_0 \frac{1}{2J} \frac{V_s E'_q}{x'_{d\Sigma}} sin(\Delta \delta) \\ -\frac{1}{T'_d} E'_q + \frac{1}{T_{do}} \frac{x_d - x'_d}{x'_{d\Sigma}} V_s cos(\Delta \delta) \\ g(x) = \begin{pmatrix} 0 & 0 & \frac{1}{T_{do}} \end{pmatrix}^T \end{pmatrix}$$

while the system's output is





 $y = h(x) = \delta - \delta_0$ 

#### 6. Case studies on robotic and electric power systems

#### 6.2 The model of distributed power generators

The interconnection between distributed power generators results into a multi-area multi-machine power system model



The dynamic model of a power system that comprises n-interconnected power generators is

$$\begin{aligned} \dot{\delta}_{i} &= \omega_{i} - \omega_{0} \\ \dot{\omega}_{i} &= -\frac{D_{i}}{2J_{i}} (\omega_{i} - \omega_{0}) + \omega_{0} \frac{P_{m_{i}}}{2J_{i}} - \\ -\omega_{0} \frac{1}{2J_{i}} [G_{ii} E_{qi}^{'2} + E_{qi}^{'} \sum_{j=1, j \neq i}^{n} E_{qj}^{'} G_{ij} sin(\delta_{i} - \delta_{j} - \alpha_{ij})] \\ \dot{E}_{q_{i}}^{'} &= -\frac{1}{T_{d_{i}}^{'}} E_{q_{i}}^{'} + \frac{1}{T_{d_{o_{i}}}} \frac{x_{d_{i}} - x_{d_{i}}}{x_{d_{\Sigma_{i}}}^{'}} V_{s_{i}} cos(\Delta \delta_{i}) + \frac{1}{T_{d_{o_{i}}}} E_{f_{i}} \end{aligned}$$

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#### 6.2 The model of distributed power generators

The active power associated with the i-th power generator is given by:

$$P_{e_{i}} = G_{ii} E_{qi}^{'2} + E_{qi}^{'} \sum_{j=1, j \neq i}^{n} E_{qj}^{'} G_{ij} sin(\delta_{i} - \delta_{j} - \alpha_{ij})$$



The state vector of the distributed power system is given by  $x = [x^1, x^2, \dots, x^n]^T$ where  $x^i = [x_1^i, x_2^i, x_3^i]^T$  with  $x_1^i = \Delta \delta_i$   $x_2^i = \Delta \omega_i$  and  $x_3^i = E'_{qi}$   $i = 1, 2, \dots, n$ 

Next, differential flatness is proven for the model of the stand-alone synchronous generator.

In state-space form one has:

$$\dot{x}_1 = x_2$$
  
$$\dot{x}_2 = -\frac{D}{2J}x_2 + \omega_0 \frac{P_m}{2J} - \frac{\omega_0}{2J} \frac{V_s}{x'_{d\Sigma}} x_3 sin(x_1)$$
  
$$\dot{x}_3 = -\frac{1}{T'_d}x_3 + \frac{1}{T_{do}} \frac{x_d - x'_d}{x'_{d\Sigma}} V_s cos(x_1) + \frac{1}{T_{do}} u$$

The flat output is taken to be  $y = x_1$ 

It holds that  $x_1 = y$   $x_2 = \dot{y}$  and for  $x_1 \neq \pm n\pi$ ,

$$x_3 = \frac{\omega_0 \frac{P_m}{2J} - \ddot{y} - \frac{D}{2J}\dot{y}}{\frac{\omega_0}{2J} \frac{V_s}{x'_{d\Sigma}} \sin(y)}, \text{ or } x_3 = f_a(y, \dot{y}, \ddot{y})$$



#### 6.2 The model of distributed power generators

while for the **generator's control input** one has

$$u = T_{do}[\dot{x}_3 + \frac{1}{T'_d} x_3 \frac{1}{T_{do}} \frac{x_d - x'_d}{x'_{d\Sigma}} V_s cos(x_1)], \text{ or } u = f_b(y, \dot{y}, \ddot{y})$$



Consequently, all state variables and the control input of the synchronous generator are written as **differential functions** of the flat output and thus the differential flatness of the model is confirmed.

By defining the **new state variables**  $y_1 = y, y_2 = \dot{y}, y_3 = \ddot{y}$ 

the generator's model is transformed into the **canonical (Brunovsky) form**:

$$\begin{pmatrix} \dot{y}_1 \\ \dot{y}_2 \\ \dot{y}_3 \end{pmatrix} = \begin{pmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ 0 & 0 & 0 \end{pmatrix} \begin{pmatrix} y_1 \\ y_2 \\ y_3 \end{pmatrix} + \begin{pmatrix} 0 \\ 0 \\ 1 \end{pmatrix} v$$



 $v = f_c(y, \dot{y}, \ddot{y}) + g_c(y, \dot{y}, \ddot{y})u$ with where

$$\begin{aligned} f_{c}(y,\dot{y},\ddot{y}) &= (\frac{D}{2J})^{2}\dot{y} - \omega_{0}\frac{D}{2J}\frac{P_{m}}{2J} + \omega_{0}\frac{D}{(2J)^{2}}\frac{V_{s}}{x'_{d\Sigma}}x_{3}sin(\dot{y}) + \\ &+ \frac{\omega_{0}}{2J}\frac{V_{s}}{x'_{d\Sigma}}\frac{1}{T'_{d}}x_{3}sin(y) - \frac{\omega_{0}}{2J}\frac{V_{s}}{x'_{d\Sigma}}\frac{1}{T_{do}}\frac{x_{d}-x'_{d}}{x'_{d\Sigma}}V_{s}cos(y)sin(y) - \\ &- \frac{\omega_{0}}{2J}\frac{V_{s}}{x'_{d\Sigma}}x_{3}cos(y)\dot{y} \end{aligned} \qquad \text{and} \quad g_{c}(y,\dot{y},\ddot{y}) = -\frac{\omega_{0}}{2J}\frac{1}{T_{do}}\frac{V_{s}}{x'_{d\Sigma}}sin(y) \\ &\text{and} \quad g_{c}(y,\dot{y},\ddot{y}) = -\frac{\omega_{0}}{2J}\frac{1}{T_{do}}\frac{V_{s}}{x'_{d\Sigma}}sin(y) \\ &- \frac{\omega_{0}}{2J}\frac{V_{s}}{x'_{d\Sigma}}x_{3}cos(y)\dot{y} \end{aligned}$$

#### 6.2 The model of distributed power generators

# Differential flatness can be also proven for the model of the n-interconnected power generators

The **flat output** is taken to be the vector of the turn angles of the n-power generators

$$y = [y_1^1, y_1^2, \cdots, y_1^n]$$
 or  $y = \Delta \delta^1, \Delta \delta^2, \cdots, \Delta \delta^n$ 

For the n-machines power generation system it holds

$$x_1^1 = y^1, x_1^2 = y^2, x_1^3 = y^3, \dots, x_1^n = y^n$$





$$x_2^1 = \Delta \omega^1 = \dot{y}^1, \ x_2^2 = \Delta \omega^2 = \dot{y}^2, \ x_2^3 = \Delta \omega^3 = \dot{y}^3, \ \cdots, \ x_2^n = \Delta \omega^n = \dot{y}^n$$

Moreover, it holds

$$\dot{x}_{2}^{i} = -\frac{D_{i}}{2J_{i}}x_{2}^{i} + \frac{\omega_{0}}{2J_{i}}P_{mi} - \frac{\omega_{0}}{2J_{i}}[G_{ii}x_{3}^{i^{2}} + x_{3}^{i}\sum_{j=1, j\neq i}^{n}[x_{3}^{j}G_{ij}sin(x_{1}^{i} - x_{1}^{j} - \alpha_{ij})]$$

or using the flat outputs notation

$$\ddot{y}^{i} = -\frac{D_{i}}{2J_{i}}\dot{y}^{i} + \frac{\omega_{0}}{2J_{i}}P_{mi} - \frac{\omega_{0}}{2J_{i}}[G_{ii}x_{3}^{i^{2}} + x_{3}^{i}\sum_{j=1, j\neq i}^{n}[x_{3}^{j}G_{ij}sin(y^{i} - y^{j} - \alpha_{ij})]$$



#### 6.2 The model of distributed power generators

The **external mechanical torque**  $P_{mi}$  is considered to be a piecewise constant variable

 $x_{3}^{i} = f_{x_{2}}(y^{1}, y^{2}, \cdots, y^{n})$ 

From Eq. (4) and for one  $i = 1, 2, \dots, n$  has a system of n equations which can be solved with respect to the variables  $x_3^i, i = 1, 2, \dots, n$ 

Actually, all variables  $x_3^i$ , can be expressed as differential functions of the flat outputs

$$y^i, i=1,2,\cdots,n$$

and thus one has

Moreover, from

$$\dot{E}_{q_i} = -\frac{1}{T_{d_i}} E'_{q_i} + \frac{1}{T_{d_{o_i}}} \frac{x_{d_i} - x'_{d_i}}{x_{d_{\Sigma_i}}} V_{s_i} \cos(\Delta \delta_i) + \frac{1}{T_{d_{o_i}}} E_{f_i}$$

one can demonstrate that the control inputs  $u_i = E_{f_i}$  can be expressed as **differential** functions of the flat outputs  $y^i$ ,  $i = 1, 2, \dots, n$ 

Consequently, all state variables and the control inputs of the distributed power system can be expressed as differential functions of the flat outputs, and **the system is a differentially flat one.** 





#### 6.2 The model of distributed power generators

Next, the **external mechanical torque**  $P_{mi}$  is considered to be time-varying

The effect of this torque is viewed as a **disturbance** to each power generator

In such a case for a model of *n*=2 interconnected generators one obtains the **input-output linearized dynamics** 

$$\dot{z}_3^i = a^i(x) + b_1{}^i g_1 u_1 + b_2{}^i g_2 u_2 + \tilde{d}^i$$
 where  $z_3^i = \overset{.}{\delta} = a^i$ 

and

$$\begin{aligned} a^{i} &= \left(\frac{D_{i}}{2J_{i}}\right)^{2} x_{2}^{i} + \frac{D_{i}\omega_{0}}{(2J_{i})^{2}} [G_{ii}x_{3}^{i}{}^{2} + x_{3}^{i}\sum_{j=1, j\neq i}^{n} x_{3}^{j}G_{ij}sin(x_{1}^{i} - x_{1}^{j} - \alpha_{ij})] - \\ &- \frac{\omega_{0}}{2J_{i}} [G_{ii}x_{3}^{i} + \sum_{j=1, j\neq i}^{n} x_{3}^{j}G_{ij}sin(x_{1}^{i} - x_{1}^{j} - \alpha_{ij})(-\frac{1}{T_{d_{i}}^{'}}x_{3}^{i} + (\frac{1}{T_{d_{oi}}}\frac{x_{d_{i}} - x_{d_{i}}^{'}}{x_{d\Sigma_{i}}^{'}}V_{s_{i}}cos(x_{1}^{i}))] - \\ &- \frac{\omega_{0}}{2J_{i}}x_{3}^{i}\sum_{j=1, j\neq i}^{n}G_{ij}sin(x_{1}^{i} - x_{1}^{j} - \alpha_{ij})(-\frac{1}{T_{d_{i}}^{'}}x_{3}^{i} + (\frac{1}{T_{d_{oi}}}\frac{x_{d_{i}} - x_{d_{i}}^{'}}{x_{d\Sigma_{i}}^{'}}V_{s_{i}}cos(x_{1}^{i})) - \\ &- \frac{\omega_{0}}{2J_{i}}x_{3}^{i}\sum_{j=1, j\neq i}^{n}x_{3}^{j}G_{ij}cos(x_{1}^{i} - x_{1}^{j} - \alpha_{ij})x_{2}^{i}\frac{\omega_{0}}{2J_{i}}x_{3}^{i}\sum_{j=1, j\neq i}^{n}x_{3}^{j}G_{ij}cos(x_{1}^{i} - x_{1}^{j} - \alpha_{ij})x_{2}^{i}\frac{\omega$$

and

$$b_{1}^{i} = -\frac{\omega_{0}}{2J_{i}} [2G_{ii}x_{3}^{i} + \sum_{j=1, j\neq i}^{n} x_{3}^{j}G_{ij}sin(x_{1}^{i} - x_{1}^{j} - \alpha_{ij})]\frac{1}{T_{d_{o\,i}}}$$
  
$$b_{2}^{i} = -\frac{\omega_{0}}{2J_{i}}G_{i2}sin(x_{1}^{i} - x_{1}^{2} - \alpha_{i2})\frac{1}{T_{d_{o\,2}}}$$

 $\tilde{d}^i = -\frac{D_i\omega_0}{2J_i^2}P^i_m + \frac{\omega_0}{2J_i}\dot{P}^i_m$ 

while





#### 6.2 The model of distributed power generators

For the **two interconnected generators** (i=1,2) one has the linearized dynamics

$$\begin{aligned} \dot{z}_{1}^{i} &= z_{2}^{i} \\ \dot{z}_{2}^{i} &= z_{3}^{i} \\ \dot{z}_{3}^{i} &= a^{i}(x) + b_{1}{}^{i}g_{1}u_{1} + b_{2}{}^{i}g_{2}u_{2} + \tilde{d}^{i} \\ \end{aligned}$$
It is used that
$$\begin{aligned} \dot{z}_{3}^{1} &= a^{1}(x) + b_{1}{}^{1}g_{1}u_{1} + b_{2}{}^{1}g_{2}u_{2} + \tilde{d}^{1} \\ \dot{z}_{3}^{2} &= a^{2}(x) + b_{1}{}^{2}g_{1}u_{1} + b_{2}{}^{2}g_{2}u_{2} + \tilde{d}^{2} \\ \end{aligned}$$
or in matrix form
$$\begin{aligned} \dot{z}_{3} &= f_{a}(x) + Mu + \tilde{d} \end{aligned}$$

It is

$$z_3 = [z_3^1, z_3^2]^T$$
,  $u = [u_1, u_2]^T$  and  $\tilde{d} = [\tilde{d}_1, \tilde{d}_2]^T$ 

$$f_a(x) = \begin{pmatrix} a^1(x) \\ a^2(x) \end{pmatrix}, \quad M = \begin{pmatrix} b_1^1 g_1 & b_2^1 g_2 \\ b_1^2 g_1 & b_2^2 g_2 \end{pmatrix}$$

Setting,  $v = f_a(x) + Mu + \tilde{d}$  one obtains

$$\begin{pmatrix} \dot{z}_1^i \\ \dot{z}_2^i \\ \dot{z}_3^i \end{pmatrix} = \begin{pmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ 0 & 0 & 0 \end{pmatrix} \begin{pmatrix} z_1^i \\ z_2^i \\ z_3^i \end{pmatrix} + \begin{pmatrix} 0 \\ 0 \\ 1 \end{pmatrix} (v^i + \tilde{d}^i)$$





#### 6. Case studies on robotic and electric power systems

#### 6.2 The model of distributed power generators

For the model of the 2-area distributed power generation system it holds that

$$\begin{aligned} x_{1,1}^{(3)} &= f_1(x,t) + g_1(x,t)u + d_1 \\ x_{1,2}^{(3)} &= f_2(x,t) + g_2(x,t)u + d_2 \\ & \cdot \\ x_1 &= x_{1,1}, \ x_2 &= x_{1,1}, \ x_3 &= x_{1,1} \end{aligned}$$

By denoting

$$x_2 = x_{2,1}, \ x_5 = x_{2,1}, \ x_6 = x_{2,1}$$





the Brunovsky (canonical form) of the distributed power system is obtained

 For the 2-area distributed power system differential flatness properties hold and one can apply the control scheme analyzed in Sections 3 and 4.

#### 7. Simulation tests

#### 7.1 Optimization-based modelling and control of a multi-DOF robotic manipulator

The dynamic model of the robot was taken to be completely unknown, while the state vector could be partially measured



7.1 Optimization-based modelling and control of a multi-DOF robotic manipulator



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7.1 Optimization-based modelling and control of a multi-DOF robotic manipulator







7.1 Optimization-based modelling and control of a multi-DOF robotic manipulator setpoint 5



#### 7. Simulation tests

7.1 Optimization-based modelling and control of a multi-DOF robotic manipulator video



#### 7. Simulation tests

#### 7.2 Optimization-based modelling and control of distributed power generators

The dynamic model of the distributed power generators was taken to be completely unknown, while the state vector could be partially measured



7.2 Optimization-based modelling and control of distributed power generators







#### 7.2 Optimization-based modelling and control of distributed power generators





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#### 7.2 Optimization-based modelling and control of distributed power generators

Table I: RMSE of the power generator's state variables					
	parameter	$\omega_1$	$\dot{\omega}_1$	$\omega_2$	$\dot{\omega}_2$
	$RMSE_1$	0.0035	0.0002	0.0034	0.0002
	$RMSE_2$	0.0123	0.0545	0.0118	0.0602
	$RMSE_3$	0.0035	0.0020	0.0035	0.0020
	$RMSE_4$	0.0031	0.0020	0.0026	0.0020
	$RMSE_5$	0.0034	0.0003	0.0033	0.0002
	$RMSE_6$	0.0035	0.0003	0.0033	0.0002



The tracking accuracy of the control method was remarkable despite the fact that

- (i) the dynamic model of the systems was completely unknown,
- (ii) only output feedback was used in the implementation of the control scheme.

It has been also confirmed that the transient characteristics of the control scheme are quite satisfactory

The proposed **optimization-based modelling and control method** is of generic use and can be applied to a **wide class of nonlinear dynamical** systems of unknown model



#### 7. Simulation tests

7.2 Optimization-based modelling and control of distributed power generators

video



#### 8. Conclusions

• A gradient-based method of assured convergence and stability has been developed. The method is suitable for modelling and optimization-based control in a wide class of nonlinear systems

• By exploiting the **differential flatness** properties of the **MIMO nonlinear model of the dynamical systems** this was transformed into the **linear canonical (Brunovsky) form.** For the latter description the design of a feedback controller was possible.

• Moreover, to cope with **unknown nonlinear terms** appearing in the new control inputs of the transformed state-space description of the systems, the use of nonlinear regressors (neurofuzzy approximators) has been proposed..

• These estimators were online trained to **identify the unknown dynamics of the system** and the associated learning procedure was determined by the requirement the **first derivative of the control loop's Lyapunov function to be a negative one**.

• The computation of the control input required the **solution of two** algebraic Riccati equation.

• Through Lyapunov stability analysis it was proven that the closed loop satisfies the H-infinity tracking performance criterion, while also an asymptotic stability condition has been formulated.





#### 8. Conclusions

• Deliverables from related research projects are:

[1] G. Rigatos, **Modelling and control for intelligent industrial systems: adaptive algorithms In Robotics and Industrial Engineering**, Springer, 2011

[2] G. Rigatos, Advanced models of Neural Networks: Nonlinear Dynamics and Stochasticity in Biological Neurons, Springer, 2013

[3] G. Rigatos, Nonlinear control and filtering using differential flatness approaches: applications to electromechanical systems, Springer 2015.

[4] G. Rigatos, **Intelligent renewable energy systems: Modelling and Control**, Springer, 2017

[5] G. Rigatos, **Journal of Intelligent Industrial Systems**, Springer, 2015



Thank you for your attention







#### Gerasimos G. Rigatos

#### Modelling and Control for Intelligent Industrial Systems

Adaptive Algorithms in Robotics and Industrial Engineering









### Gerasimos G. Rigatos

## Advanced Models of Neural Networks

Nonlinear Dynamics and Stochasticity in Biological Neurons

2 Springer









Studies in Systems, Decklos and Control - TE

Gerasimos G. Rigatos

### Nonlinear Control and Filtering Using Differential Flatness Approaches

Applications to Electromechanical Systems





